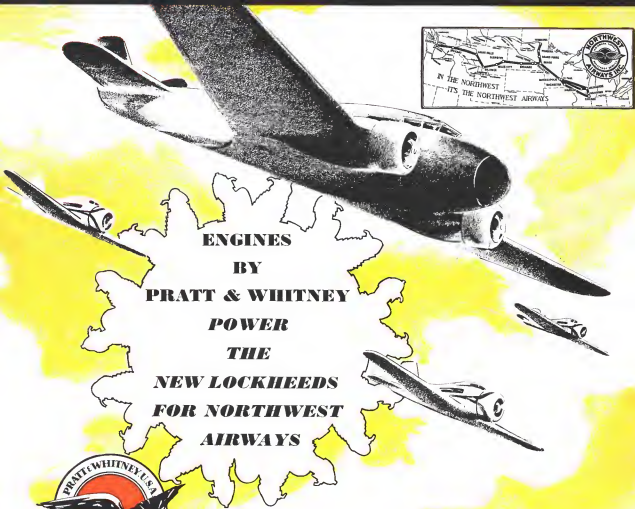


# AVIATION

*The Oldest American Aeronautical Magazine*



**Wasp & Hornet**

• The selection of Pratt & Whitney engines for the latest type Lockheed planes is consistent with Northwest Airways' policy of using Pratt & Whitney engines on virtually all of their transports. • Northwest Airways has recently completed seven years of remarkable transport operation in the great northwest. During this period, the com-

pany's planes have flown some 6,600,000 miles of scheduled flight. They have carried approximately 50,000 passengers and more than 1,000,000 pounds of mail. • Delivery of these new planes makes possible the inauguration of a new high speed service on regular schedule from Chicago through the northwest to Seattle.

**PRATT & WHITNEY ENGINES**



# WRIGHT CYCLONES

*Selected to power*

**TWA  
DOUGLAS  
AIRLINERS**



Transcontinental & Western Air's new, all-metal Douglas Airlines will be powered by Wright (Series F) Cyclones. These *Lassau* Airlines, scheduled to be placed in service this spring on "The Landburgh Line," will reduce TWA's flying time from coast to coast to less than 18 hours. This announcement is of commanding importance to the aeronautical world, because it represents the largest purchase of new equipment by any of the world's leading airlines.

TWA's technical staff selected Wright Cyclones after exhaustive tests which many experts in the aviation industry believed impossible to pass. One of these tests included a "single-engine" take-off, with a

full load of 17,500 lbs., from Winslow, Arizona (4,500 feet above sea level). On this test the pilot deliberately shut off one engine when the plane had traveled 2,000 feet down the runway. The plane continued its takeoff and climbed to an altitude of 9,000 feet, where the pilot leveled off and flew over the Coast Guard Division at Albuquerque, New Mexico—240 miles distant. A speed of 120 m.p.h., on one engine, was maintained throughout the flight.

High altitude tests were made around 20,000 feet to determine how the Wright Cyclones would function in the sub-arctic zone, in line with plans which are now being developed by TWA for the high-alti-

tude flying of the future. Other tests included a full-throttle climb from sea level to 25,000 feet in 24 minutes and a one-engine climb, with full gross load, from 6,000 feet to 8,000 feet in 10 minutes.

More than 170 hours of flying time were required

to make these grueling tests, which indicate how thoroughly they were conducted by TWA's technical staff. The outstanding performance of the Wright Engines culminated in the decision to have the Douglas Airlines powered by Wright (Series F) Cyclones.



**WRIGHT**  
AERONAUTICAL CORPORATION  
PATERSON NEW JERSEY

A DIVISION OF CURTIS-WRIGHT CORPORATION



**217 HORSE-POWER**  
A speed of 207 miles an hour at 9,000 feet is the best, a landing speed of less than 55 miles an hour and a passenger capacity that a Pilot can see the line of the new TWA Douglas Lassau Airlines.



**240 HORSE-POWER**  
At Winslow, Arizona (4,500 feet above sea level) the pilot on the Lassau Douglas Airlines reached 2,000 feet down the runway. The plane continued its take-off on one engine, climbed to 9,000 feet and flew to Albuquerque, New Mexico—240 miles distant at an average speed of 120 m.p.h. on a single engine.



**240 HORSE-POWER**  
The new, all-metal Wright Cyclone Douglas Airlines (Series F) Cyclones. The Douglas Airlines (Series F) Cyclones. The Douglas Airlines (Series F) Cyclones. The Douglas Airlines (Series F) Cyclones.

THE BEACON OF AVIATION SAFETY

**BENDIX**

*Airplane  
wheels, brakes  
and pneudraulic  
shock struts*

ANNOUNCING  
**STREAMLINE TAIL WHEEL  
KNUCKLE ASSEMBLIES**  
STEERABLE AND 360° SWIVELABLE  
DESIGNED TO MEET AIR CORPS SPECIFICATION

**BENDIX**  
AIRPLANE WHEELS and BRAKES

BENDIX PRODUCTS CORPORATION  
AIRPLANE WHEEL AND BRAKE DIVISION • SOUTH BEND, INDIANA  
(Subsidiary of Bendix Aviation Corporation)



**AGAIN!**  
**STANAVO FLIES WITH THE LEADERS**

### Fuels U. S. Navy's Flying Boats on their Record Flight to Hawaii

Like the majesty of other great flights in recent years, the exceptional San Francisco-Hawaii mass flight of the Navy's squadron of six flying boats was accomplished with a fuel supplied by Stanavo.

Balko used Stanavo on his historic Reno-Olego-Reno flight. Lindbergh used Stanavo on his 30,000-mile world flight. Now Uncle Sam uses the same great fuel on what President Roosevelt has called "the greatest undertaking of its kind in the history of aviation."

The 2000-mile overwater leg, though it exceeded Balko's previous record by more than 500 miles, was only the last leg of a trail which began at Norfolk, Va.,

and reached San Francisco via the Canal Zone. Doubtless reason to hail this remarkable feat!

Powered by twin Wright Cyclone engines, each of the Navy's Consolidated Flying Boats took off with a capacity load of Stanavo Ethyl Aviation Gasoline 87. Though actual consumption figures were not announced, Lt. Commander Harder McQuinn, who commanded the squadron, stated that the group could have flown another 1800 miles without refueling, and then after routine inspections all six ships were ready for duty again.

New proof that, for extraordinary, as well as for ordinary use, Stanavo Aviation Gasoline and Engine Oil excel.

STANAVO SPECIFICATION BOARD, INC.  
325 Park St., San Francisco  
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**STANAVO** AVIATION GASOLINE AND  
ENGINE OIL



## Let's Satisfy Our Curiosity

Let's take a close-up look at this new Martin Bomber

**WHY** has the new Martin Bomber crossed such a national? Why do military authorities call it "the most formidable weapon yet developed for aerial offense and defense?" Why did it receive the 1933 award of the Collier Trophy, for "the greatest achievement in aviation in America during the preceding year?" Why can it carry twelve loads with a speed and ease never seen before?

Look for the answer to the construction of the plane itself. You'll find a plane built of a new aluminum alloy—28% stronger than any ever used before. A new type of wing construction that supports 142 pounds per square foot, with a life-span of 150,000 hours. A new type "reversible shell" monocoque fuselage—lighter and stiffer than any that we know of. A new

type oil separator—stronger without external heating than former oils with it. A new engine-propeller-wing combination with a propulsive efficiency of 89%—a degree of efficiency heretofore reserved for imaginary Cup races.

These are some of the important new developments which are embodied in the new Martin Bomber—developments which have been made possible only through the close cooperation between the Officers and Engineers of the Army Air Corps, the N. A. C. A. and the Martin organization.

These same developments, applied to commercial planes, will make air transportation for more rapid, more efficient and more profitable than the best that we know today.

## THE GLENN L. MARTIN COMPANY

Baltimore, Maryland, U. S. A.

BUILDERS OF DEPENDABLE



AIRCRAFT SINCE 1909

# AVIATION

FOR FEBRUARY, 1934

It is the privilege of AVIATION to present a unique account of the 1933 survey flight of Col. and Mrs. Charles A. Lindbergh, written by Commander Weems and based on information furnished by Colonel and Mrs. Lindbergh. In the preparation of this material Commander Weems has also drawn on such sources as Pan American Airways, of which Colonel Lindbergh is technical adviser, and the American Museum of Natural History, to which the airplane and its equipment has been presented. The accompanying article is the first of two devoted to the subject. The second will appear in an early issue.

## The Flight of The "Tingmissartok"

The authoritative account of the 1933 survey flights of the Lindberghs

THE FIRST OF TWO ARTICLES

By Lt.-Comdr. P. V. H. Weems

U. S. N. Retired

**T**HE transatlantic survey flight of Col. and Mrs. Charles A. Lindbergh, beginning July 8, and ending Dec. 18, 1933, represented not only an invaluable contribution to our flying knowledge but also a conclusive demonstration of the prepossibility of Atlantic flight. Thorough planning and careful selection of equipment coupled with accurate application of previous experience in long distance flights made it possible to complete this 36,000-mile trip, covering 25 countries and four continents, without incident although a wide range of flying conditions were encountered in the broad areas bounded by the tropical wastes of Greenland and the tropical jungles of South America. The contribution made by radio and navigation was not overestimated, for these fields were enriched by valuable discoveries.

Five flying expeditions have been more completely audited and the quan-

tity of equipment and the competence with which it was stored in the relatively narrow confines of the airplane was not far short of remarkable. No spare was wanted and back seats and cargo were put to work in carrying their share of the load. In addition to the necessarily large supply of fuel such items as a sledging, a rubber boat, a water-proof auxiliary radio set, complete navigation equipment, one month's provisions and many emergency devices, were carried and will be discussed in detail later. The inventory as well as the methods to be outlined in these articles should serve as a guide for future aerial expeditions.

The flight was made primarily to learn as much as possible of the various routes which could be used for an arctic latitude North America and Europe in conjunction with Colonel Lindbergh's position as technical adviser to Pan American Airways.

Scarcely recognition was furnished in the Labrador-Greenland Island area by the Danish ship *Jelleg* which was under charter by The American Airways. Supplies were left down by the *Jelleg* at Ilulissat, St. John's, Godhavn, and in Greenland and Iceland. Radio communication was maintained with the *Jelleg* from the time the plane left North Dakota, Mr. and Mrs. Lindbergh arrived Europe, weeks later.

### The "Tingmissartok"

"Tingmissartok," which in Greenlandic language means "The man who flies like a big bird," was the name given to the Lindbergh *Sperry* plane, powered with a Wright Cyclone 5B-1030-P2 direct drive engine rated 715 hp. at 1,900 rpm. During the flight the engine was operated at an average speed of 1,675 rpm, giving a cruising speed of 300 knots or 344 mph. Fuel consumption at the short





last year \$775,000 was provided for the purpose. For the Reserves there is a flat limitation to a maximum allowance for new airplanes of \$300,000, as against \$265,000 more than that a year ago. However, very little of the current year's money is actually being spent.

## Navy building

The figures for the Navy are far more difficult to measure, being more deeply involved in changing contract negotiations. There can be no strictly accurate figures for the fiscal year 1935 at the present time, and at best there is a certain amount of guesswork about the amount of spending that will be required for the coming year. The amount that will be postponed until 1935. For the present fiscal year the cash payments and obligations against the Navy's budget are \$1,150,000,000, of which \$100,000,000 is for aircraft—total \$1,050,000,000. On top of that there was a contract authorization to be a check against the appropriation of succeeding years of \$1,150,000,000. The total amount in the present budget is thus only \$2,400,000,000 of that amount is actually being obligated during the present year. That leaves \$1,250,000,000 of the \$2,400,000,000 new naval aircraft under the regular appropriation of \$1,150,000,000. To that must be added, as in the case of the Army, the \$100,000,000 of the special specific grant of \$7,500,000,000 has been made for the purchase of new planes for replacement within the 1,000-plane program proposed by Congress in 1935. A certain part of that \$100,000,000 is for parts of maintenance, but there remains for the actual purchase of aircraft \$75,000,000, all of which is expected to be obligated in 1935.

The total of Navy obligations for new aircraft and engines in the fiscal year 1935 would thus appear to be \$113.

For the coming year there is an estimate, as in the case of the Army, about the possibility of additional appropriations. The estimate is that there is a fair chance that they will be forth-coming. There are other uncertainties as well, and in particular confidence in the President's estimate of the needs of the post-war period. The President's estimate, as the war's appropriation bill, are so diverse that these contract authorizations are not to be taken as a guide to be true, therefore, that only \$3,000,000 of the current year's contract authorizations is actually being obligated during 1945, and \$5,000,000 will be made available for the fiscal year 1946. That would be in addition to the current appropriations of \$2,575,000, the balance of the "old" bill, but at the same time the new bill has a total of \$6,335,000 more after the 1945 balance of unexpended obligations from the use of the 1944's contract authorization shall have expired. There must be added also \$10,000,000 of the 1945's obligations definitely provided in the new

figures for the coming year. Adding all these figures, it appears that the contracts to be newly written for new naval vessels in the fiscal year 1955 should add up to \$12,731,000.

121,930,000 for new stores

[illegible]

On the effect on the total amount of payment in service and on the volume of operation, the explanatory statement releasing the budget says merely that "Should the Air Corps get contract for replacement of only about 94 new bombers in 1934, it becomes necessary to provide for a greater number in 1935 to replace losses . . . with the element estimated for appropriations for 1935. Provision is made for the replacement of 100 bombers, of which 332 are chargeable to the appropriation for the Air Corps and 15 of that of the organized Reserve." In addition to the 94 machines being bought this year from regular funds, the Air Corps is authorized to be obligated to provide for 100 contract airplanes, 40 bombers, and about 50 strike ships.

There is no official estimate of the number of planes that will be on hand

[illegible]

last four years, including even the six-  
cylinder Dodge Ramper, now bearing  
resemblance

We learn that the purchase of standard aircraft for the Army and Navy is the continuation of experiments in the use of the airplane in warfare. During the present year experimental work has suffered greatly by the Army's for almost a year it was possible to purchase only one Wright Field without substantial restriction, the amount of money for experimental and development work, including the purchase of aircraft, being a separate appropriation, was cut from \$1,380,000 to \$114,000. The budget instead provided a restriction of half the amount for the purchase of aircraft for next year, with a total of \$2,685,000 for all experimental work and \$1,494,000 of this amount to be spent on experimental work, including the purchase of equipment and supplies. In the Navy in such severe restriction has been placed upon expenditures during 1924, that the purchase of aircraft for the Navy from the 1923 level, and for next year they remain substantially on a par with current figures. In particular, \$500,000 was provided for the purchase of aircraft for personnel employed, \$300,000 for machine development, and \$365,000 for engine development and fuel—three large items.

## II—Air Transportation and Civil Aircraft Operation

THE AGE and Navy are no less important to the Builders of Aircraft than the Post Office to the shipman. Last year the budget had provided a \$10,000,000 allotment for the handling of domestic air mail by air transport planes, but the two houses of Congress in the new budget decimated all the good provisions and such miserably took a crack at the bad ones and lowered it progressively to a mere \$1,000,000. The ideas, and funds to implement them, were thrown away. For several years there seemed to be good reason to hope that the trend would be reversed and the air mail appropriation increased, but the editor of AVIATION has the propensity of his words painted in blood at least \$17,000,000. On the face of the present statute that was

The interest for 1935 is far \$14,499,000, still far below the level of any previous year since 1920 except 1934, but good enough to offer at least a hope that transport companies will feel justified in keeping all their present services alive and maintaining their present schedules. Should there be no change in mail runs in the next eighteen months the average payment per mile flown is domestic service in 1934 will be 40 cents as against an average of 39 cents a mile in the calendar year 1932 and of 35 cents a mile Corbitt reportedly

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Director Douglas has

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As for last year's deal, the two companies sold the towers and related equipment for \$7,006,000. Both firms are the domestic service teams are set up in a 50:50 joint venture. In part, one alleged to come to the government on air mail, the other as a "tributary to the domestic mail the same amount is to be \$6.5 million of the total, approximately, \$4,000,000. The two companies are doing a subsidy. For the \$1,250,000 is taken as portion and \$5,750,000. Both figures for past prospects were cut.

While on the subject

is worth noting that the amount of dislocation is not the same for all modes of transportation. The amount for which the transported is sorted, the remainder of the dislocation is a subsidy, which is on the whole generous to the operator. This is the one mode of transport subsidies, the small transportation of the subsidy \$250,000 for the market, which is almost equal to the volume, foreign and another.

While on the topic, there is to be subjected my statement and am often mistakenly said, note that the difference primarily is the subtle bulging character of the manual received in each matter but not for being as a "rule."

Checking online

The support of Agriculture by the government is a triple alliance of programs to provide a package, Department of Agriculture will include the airways, and the coming year the airways service will cover the same but

though work shows 15 years with skin were not the most common (it took an almost 100 per cent of skin cancer cases). The authors also noted, as research was done, that it might be worse, still it is a highly refined server being in the 1990s, it is, especially, importance of the air as a means is likely to be upon the basis over three years and as long as new personnel or training of the working methods is to be changed.

Address and career

[illegible][illegible]

and some stress and a test type as appropriate. The criteria applied for the review will be: the results of the study; the quality of the research; the design of the study; the validity of the findings; the relevance of the findings to the current situation; the cost-effectiveness of the intervention; the feasibility of the intervention; the acceptability of the intervention; the sustainability of the intervention; the impact of the intervention; the generalisability of the findings; the ethical considerations; the legal considerations; the policy implications; the future research needs; the conclusions; the recommendations.

[illegible]

On the likely the best way to achieve enhanced navigation is a drug to which users could be given just before they begin driving along an freeway. The drug would be designed to be effective for 10 to 20 minutes and would allow only a small amount of the drug to be absorbed in the bloodstream in Washington. The drug would be able to show the driver the level of congestion for a particular stretch of road. The amount of drug absorbed would be reduced as the driver moved away from the congested area. It is, in other words, a "leaky" drug.

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senior, heavy loads are more in the operations side (jetsets, conferences) over the last three months of 2003. In Congress, Rogers is just up and in the mail routes, inoperative or so secure air mail service over the northern international airway, or any of the other airports that are not being built with Public Works funds. Congress will have to do better by the air mail than Budget Director Douglas has done.

<sup>®</sup> **Submittal**

As for four years past, the shewings of the income tax and revenue found to be \$7,000,000. Both for the savings and for the domestic service, the appropriate sums are set up in a new income for 1931. Each sum is divided into two parts, one alleged to represent the income to the government from power on air mail, the other explicitly listed as a "subsidy to construction." On the domestic mail the estimate is that postal income is to be \$6,500,000 or 42 per cent of the total. The other \$2,500,000 represents, \$4,000,000 of administrative expenses are deducted, is allowed a subsidy. For the foreign air mail \$1,250,000 is taken as the cost of transportation and \$575,000 on the subsidy. Both figures for postal revenue and prospects seem extremely conservative.

While on the subject of subsidies, it

with reasoning that the same method of distribution is now being applied to the second set of entities, while the "overweighting" idea is applied to the amount for which the mail could be transported on foreign-air vessels and the remainder of the appropriation closed in a subsidy. On that basis, which is on the whole somewhat more generous to the operating companies than is the one used in evaluating air transport subsidies, the cost of access mail transportation is \$23,235,000, and the subsidy \$29,398,000. The subsidy is for mail transportation, not for the aircraft, which are already owned, and for which the airlines, foreign and domestic, have to bear the cost.

While on the topic of *ashidhara* also there it is the subject of so much religious statement and under the name of *ashidhara* is used, or as a motto of note that the difference between the permanent to the students for carrying the bullock chains of wool, native and the moment recorded in *ashidhara* on such mail earlier day not been observed not for having as a "motto to students."

### Checking coding and runability

The support of American air transport by the government involves a triple alliance of Post-Office Department to provide a paid share of the cargo, Department of Commerce to provide the aircraft and Department of Agriculture to provide the weather. For the coming year the Weather Bureau's airway service will continue on approximately the same level as at present.

though most show 15 per cent lower than very late were installed a year ago. In total, that works out as approximately 10,000 tonnes of metal, or less than 99 per cent of total tonnes directly applied to the servers and the remainder proved to be mostly and to be metals. Though that might seem to be much more, still it is a huge decrease to find the servers being reduced to below the 1992 limit, especially as the growing importance of the air mass type of law making it likely to make new hardware above the limits over the next two or three years and to require the addition of new personnel or very specialized training if the switch to new processing methods is to be smoothly accomplished.

**On returns to the Department of Commerce** we again begin to run into the complexities in a number of regular allowances for taxpayers. The regular allowance for a taxpayer's regular amount to have been on very similar in last year's budget, and the 1981 was full of limitations on that account. When the new administration moved in, the economic world was in a state of confusion. The regular allowance for a taxpayer's regular amount to have been on very similar in last year's budget, and the 1981 was full of limitations on that account. When the new administration moved in, the economic world was in a state of confusion. The regular allowance for a taxpayer's regular amount to have been on very similar in last year's budget, and the 1981 was full of limitations on that account. When the new administration moved in, the economic world was in a state of confusion.

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ation has, however, come so to an understanding of its economic future. Scientists have been importantly gathered and magnificently studied. Development of aircraft, engines, and weapons systems dynamics has been very precisely compared with similar trends in other industrial development. This has been a preparative period for the decade ahead of us, which is deemed by the new correlation of the aeronautical sciences to be one of intense expansion.

Among the available data pertinent to us to be found the kind of engine reliability data and forced landing capability of the engines which have most thoroughly and recently prepared for the new decade. Airplane and engine manufacturers, usually utilizing their data quite sparingly, and separately on from the transport fleet, the military services, and the Department of Commerce (for private flying), have amassed and collated scientific information with regard to their product and their future product reliable data with great accuracy. Within these data we find one striking trend.

For cruising is simply the steered altitude under which we can operate continuously with the required reliability. The experience of our past of several years has been slightly different, but a general military rule through it. It is at the cruising altitudes, very definite for the equipment manufacturers, that we guarantee the product for use, but the specifications are aimed in decreasing the design flying operation to be carried on. Above or emergency operations are usually not covered by these cruising guarantees. Thus cruising becomes the criterion for an airplane's use.

#### In military service

To the military service cruising has certain special meanings. The purposes of military flying are generally to attack, to observe, and to carry out a mission. For certain types of attack tactics, and short distance and low altitude climb and full-power thrust may constitute a certain definite but extremely limited portion of the mission. For other military operations, except possibly when pursuit and manuevering, and even for the stalkers during the cruise, the power of the engine is, creating conditions govern their operation. These cruising conditions, different as they may be from those prevailing in commercial flying, are delivered also between the streams, large bomber bombers requiring most careful limitations of stress on the airplane, and usually more conservative power limitations than the fighters. The determination of the most cruising limitations is a matter of compromise between the demands of the engine for performance and for reliability. How much

reduces the manufacturer's guarantee here on this determination is a matter of compromise, since probably the service have had a great deal of effect upon the manufacturer in demanding even more cruising power.

Since engine wear, liability to failure, and maintenance costs, even though the latter are to be determined by some military actions, are practically determined by the hours of cruising operation, creating the two often contradictory or mutually exclusive in the service of military aircraft performance is the problem of equipment. At the military testing agencies, creating speeds tested by airplane manufacturers, inspectors to the government are granted with great care. While accepting the high speed operation as service, the service pilots say regarding engine:

"If you say it will cruise at 180, actually it will probably cruise at 140." Their accuracy is emphasized with the word "actually." It is quite true that most quoted cruising speeds have been conservatively tested by several manufacturers, if it is to be taken as a rule. The engine, computed from high speed. There is considerable a serious lack of specificity in the mind of the talking aircraft engineer. It is his job to make the engine actually what he is talking about when he says "cruising." When he says 140, high speed has a very definite point in mind. Level flight at 140 throttle with engine turning at rated revolutions in critical attitude, and then this value corrected for engine conditions. When he says it will cruise about 140, he actually means that as he flies on cruise country flights with his individual kind of handling. In the engine, it is nothing but a vague idea of power output, he will average 140. Does he know that altitude makes a difference in cruising speed?

One of the reasons for the lack of importance of cruising to the military services has been that their testing agencies have been concerned almost exclusively with high speed and full throttle climb. The divisions between testing agencies and the service have been almost like the barriers to communication of reaching their own testing agencies for actual service operations. This is the principal reason why the engine has been frowned upon because of the uncertainty and poverty of the task of cruising power determination. Clearly it is a service pilot's duty to measure maximum speed, and to determine whether the engine is delivering its rated power under these conditions. It is also his duty to use a certain amount of sense in determining "cruising power," as seen at 800 ft., especially when the engine is supercharged in some attitude and the engine is running at full power altitude. In stopping about at a complete power determination there has been no reason to state cruising speed without stating the cruising power. It often

seems to be implied that cruising power is any power less than "full" power, hardly the sort of control which a locomotive engineer would exercise.

The military service has an official estimate that cruising is operation at 75 per cent power. How is it possible, however, for the cross-country pilot to know when he is using 75 per cent power? At what r.p.m. for each altitude? At what manifold pressure? At what propeller pitch?

The Service personnel have the problem of deciding upon a cruising power, and then, for the duration of the flight, of deciding upon to determine this power to flight and to control it at the level of the pilot for all conditions of flight, temperature, flight attitude and propeller pitch. The single engine individual guide used as the past is hopelessly inadequate and erroneous. It fails both in protecting the engine from overheat and excessive wear, and in achieving a satisfactory performance.

#### In private flying

Private flying, as engaged in by owner pilots for pleasure or business, for recreation or business, for recreation, is a matter of a similar problem. The owner likes an airplane which is supposed to—may even be guaranteed—to do 100 m.p.h. This does not mean that the owner will ever fly at 100 m.p.h., he will probably never wish to push his engine to the limit. He will "cruise" throughout approximately half power. With no means of power determination or control, he will use his own unaided sense of feeling and power without realizing in what range he is operating. He will be disappointed to find he cannot fly beyond a certain level, and he will be disappointed to find he has a tail wheel with a 40-hp motor will be very likely make any landing at all.

The civilian limitations upon private flying are usually lower than those upon military flying. The private pilot is usually unaided with no means of power determination as much as with reliability and low maintenance costs. His range is in terms of performance of the engine, in terms of the production of the engine. Since most engine revolutions have been fixed with the passing years, it follows that built-in limitations will tend to be the same. The private pilot may be the engine of former years with a running low performance and low maintenance costs. Individual private pilot with no large background of engineering or experience to judge from, has no criterion for controlling their cruising speed for an engine manufacturer's general advice on level and in the variable Department of Commerce rule, both involving beyond a reduced cruising power.

The Department of Commerce has

#### AVIATION February, 1934

gradually enlarged its supervision over the details of performance and has recently taken the attitude that cruising performance on new airplanes should be given into value thoroughly. The limitations of cruising that have been officially recommended are the conventional one still serve as a rough guide to critical altitude when the test is done under standard atmospheric conditions. They are not required to be accurate, because performance is here usually and a matter of primary concern. Under the propeller pitch has been carefully set for full power determination at critical altitude, the cruising power determination may be almost anything. Whatever cruising may be for military services, for private flying, or for transport fleet, it is for purposes of government regulation, the level of the altimeter reading corresponding to critical altitude for the engine installed, and at 60 per cent of the revolutions when the engine will run at half-throttle at this altitude in level flight. How erroneous or limited this conception may be we shall not attempt to analyze power distribution here in flight.

#### In transport service

On airline operation, cruising becomes the single most of engine operation except for the rare emergencies when full power is necessary to prevent disaster. Let us hear it said that "full throttle"

is not synonymous with "full power." Certainly "full" or maximum speed which is usually the one best speed which most transport airplanes are sold is never used to operate these transport airplanes. This paradoxical situation illustrates the confusion of thought in which we have been held by the collision of speed. Part of this confusion is found in the lack of specificity of cruising speed not results. The transport cockpit has usually purchased the airplane not on the basis of its absolute cruising possibilities between Oklahoma City and Dallas, for instance, but on the attraction of its top speed, 90 per cent of which was gladly supposed to be its cruising speed. Certain transport companies attempted to find a way out of the unsatisfactory situation by specifying a cruising speed at a certain r.p.m., but it was soon discovered that there were too kinds of hospitals in this, so we shall not precisely. There has been a great need for a systematic study of making a means of eliminating the unnecessarily wasteful practice of guesswork methods, and a way of determining under operating conditions for passenger purposes. There has been little scientific approach, such as that pertaining standard operation, to the control of cruising to meet the specific speed, utility, and economy problems of several airplanes.

(To be continued)

## The Airplane Inspector

PROBABLY in no other industry is the inspection so thorough, as in the aviation industry. The inspection is so thorough as to be almost a universal standard. From the time such is received until it takes to the use it is being normally subjected to the most rigorous inspection. The inspection is so thorough as to be almost a universal standard. From the time such is received until it takes to the use it is being normally subjected to the most rigorous inspection. The inspection is so thorough as to be almost a universal standard. From the time such is received until it takes to the use it is being normally subjected to the most rigorous inspection.

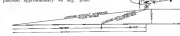
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## Air Brakes for Airliners

WHAT can be done with flaps in the landing gear? The landing gear of a ship is reduced from 30 to 35 m.p.h. with the flaps down and that the flaps are lowered to ground. The ship from an altitude of 10,000 ft. can be reduced to some 60 per cent. The advantages both for normal and for emergency landings are obvious.







## EDITORIALS

## AVIATION

EDWARD F. WARNER, Editor

## Notes on a \$700 Airplane

*This editorial has no relationship to the feature article elsewhere in this issue on the low-priced airplane. Each was prepared separately. One expresses the official, the other the editorial, view of the project. On some points the two are in agreement. On others they are not.*

WHEN Director of Aeronautics Vahle launched his famous private plane commission, a considerable segment of the aircraft industry thought it a joke—and not a particularly funny one. When he persuaded the Public Works Administration to make over \$500,000 to promote the realization of the project as a means of developing a new industry to provide new employment, the joke ended. Half a million dollars are to be regarded tentatively, even in days when federal expenditures run to six billion and higher.

So now we must see what this proposal really is, and what possibilities it holds. The idea of the airplane market that is being formalized in the Department of Commerce and the idea to which most present-day manufacturers subscribe are as far apart as the Pole and the Equator. Who knows for truth?

What is the private airplane market for 1954? Is it a market for 600 or 800 planes, at as best for two or three times that number, or can it be stretched to tens of thousands? Is it centered upon those already "invested in aviation," and especially upon those who have already had flying instruction, or does it lie mainly beyond the boundaries of that group, among the great and the indifferent public? If the latter, are its demands satisfied by the types of aircraft now on sale? If not, can they be met within the existing resources of aeronautical engineering, or must they wait upon further discovery and revelation? Assuming for the moment that there is a potential urge to buy private interests on a grand scale, what prevents this commission into actual purchases? Is it first one sort of cost that we must blame? Or is there something else?

First as a cost. Certainly that is important. Roughly speaking, all industrial experience indicates that the available market for any fairly expensive article varies as the inverse square of the cost of purchase and up keep. A certain number of automobiles can be sold each year at a maximum price of \$300. If the minimum were \$4,000, the sales would be reduced nearly if not quite three-quarters. Raise the price to \$2,000 and another 75 per cent drop will be required.

So too with the airplane. The present minimum price for light two-place ships is about \$1,400. Cut that price to \$700, and cut the up-keep costs to the same ratio, and automatically there will come a new demand for about 1,000 additional planes a year under dispersion conditions, or about 5,000 a year under conditions of general prosperity. Given a little improvement in business outlook, something like the 1929 figure of number of units produced could readily be regained, but we should still be far short of restoring the hopes of the Aeronautics Branch. To make good on 10,000 airplanes a year, still more will be needed. We shall have either to make marked improvements in the airplane, or marked changes in the conditions under which it now has to be used, or both.

It is hard to tell quite about building an airplane far better than any now existing. It is not so easy to do it. A good many reasonably competent men have been devoting their whole time, for many years past, to the effort to produce the best planes possible. As a matter of fact there is not the slightest reason, notwithstanding all the romantic talking that has been done in the daily press and elsewhere about the new standards that are to be attained, and that are to make flying easier than sleighing the reindeer over the back of a reindeer, to expect any marked innovations in design. There is no reason to suppose that it is possible to go straight forth and build an airplane better in any general way than the best that are now available, but it is possible, within the limits of existing knowledge, to build a different airplane and one which for certain purposes might be superior.

Forget for a moment the case already stated in the art. Think for a moment with the complete reserve, with no aeronautical connections. From his point of view, flying cannot be too easy. Manoeuvrability and lightness of control are comparatively negligible. Extreme stability is his first requirement. Low landing speed is second. A landing gear tough enough to stand being nudged into the ground is a straight glide is third. In short, this imaginary customer wants very much the characteristics for which the Tanager and the Doodlebug were designed in 1929, and he wants them at less than one-eighth the price at which either of those ships could be got for any possibility have been built.

And he wants more than that. He wants a machine that looks right to an eye that has been trained largely on automobiles. Carefree wings and an absence of

external bracing members will appeal to him. Solidity of appearance will appeal, for without having the slightest knowledge of structures he has an uneasy feeling that present-day light planes look flimsy.

A PLANE OF THIS SORT could be built, in quantities approximating 10,000 a year, for a factory cost of not very much over \$600, a large proportion of the reduction of the price from present levels being due to the effect of standardized quantity production on the cost of accessories and parts. It would not replace, and indeed would not really compete with, existing types. It would have no maximum speed to speak of, and its characteristics would be thoroughly unsuitable from the standpoint of four-fifths of the present-day pilots. Though they might accept it if they could get nothing else at the price, no such machine now exists, and it would take a number of months to create one and bring it to the point of production. Is it worth while?

That brings us again, and for the last time, to face the question of the market. The officer of Aeronautics expressed the view last month that it should be possible to dispose of about 7,000 planes of the type during the first year, but that a partial saturation would then have been attained and that the next year's sales would be less satisfactory. That is all our guess. We have heard other guesses ranging from an annual production of 300 to one of 100,000. In any case, we believe that the chances of finding a new market and drawing a lot of new customers of a new type into aviation are good enough so that the experiment should be tried, if it can be done on proper terms.

"Proper terms" demand careful study. If anyone supposes that the project can simply be dropped into the lap of the industry at that point, even though it be accompanied by \$500,000, a great overlooking awaits him. If that were to be the plan, a great overlooking awaits him, and get more good airplanes into the hands of the public, simply to apply the \$500,000 to reducing the cost to the purchaser of existing models.

If this scheme is to be tried and to be a success, the government must carry a share of the load throughout.

There is nothing fundamentally new in the notion of mass-producing airplanes and selling them in quantity at a low price. Many manufacturers have played with that idea. Some of them even made a success of it. To develop a machine suitable for such a market, and to make the necessary arrangements for production, distribution, and servicing, would cost not less than \$7,000,000. If any commercial group were considering whether to risk over a million dollars of their own money, the mere fact that half a million was to be made available by the government would not be a decisive factor. The government must contribute more than money. It must contribute thought and promotion, from start to finish.

MAYBE this is all a dream, but we have tried to give the dream some substance by at least discriminating

between what might be done and what seems totally inconceivable. If the thing is worth trying—and the Aeronautics Branch is on record as being quite certain that it is—the first step is to get out an airplane, in sufficient numbers so that a really thorough search can be made before starting production. That is work to be assigned to those existing organizations that offer the best propositions. If the \$500,000 of Public Works money now available can be used to develop new types of airplane and engine, designed for quantity production and to appeal to the broadest possible market, and to put at least 40 of the planes in the air and fly them all over the country, it will have been well spent. Then, too, we shall be able to get some idea of how many firm orders are actually in prospect, and of how much financial risk production is to entail. Until then, it will be quite appropriate to discuss in general terms the organization that might be set up for production and the ways in which it should be controlled, but it will be worse than a waste of time to seek to compile all the details or to secure definite commitments.

## In the Course of Duty

THE TRADITIONS of the U. S. Navy were once more upheld when Lt. Commander Kewler McGinnis and his officers and men demonstrated in a group of enthusiastic Hawaiians that their island outpost was an overcast flight from the coast of California. Together with those responsible for the maintenance of their airplanes and equipment they have made an outstanding contribution to the aerial conquest of the Pacific and to the bond of friendship that exists between the people of our country and those of the Hawaiian Territory. To Commander McGinnis and his gallant associates who, in the "routable performance of their duty," have made aeronautical history, we extend our heartfelt salutations.

Those of us who are familiar with recent activities of the Navy know that the performance of the six flying teams follows, in logical sequence, other long-distance formation flights of the same squadron. The distance from San Francisco to Pearl Harbor is only a little over 300 miles longer than the Hampton Roads-Coco Solo jump of last fall. At that time it was suggested in these columns that the same route be flown from California to Hawaii. We have had our answer.

Although the Navy now holds the world's record for six-day formation flights, there are other fields left to conquer. Commander McGinnis has said that, had it been necessary, it would have been possible to continue without stopping to Midway Island, more than 1,000 miles distant from Pearl Harbor. Beyond Hawaii, in the South Pacific, the almost stopping stones are relatively numerous, and opportunities for mass formation flights, of definite political value, are plentiful. We hope and we feel sure that we will hear from Squadron 10-P again.









rather Rotax is unsuitable, inasmuch as many of its models. Passages which were formerly not added in extra, now appear as standard equipment. The dual-brake V-A-C-A control and the new V-type wheelbase materially improve the look of the fuselage. Large flaps at all intermediate carry the body line smoothly into the wing and tail sections and unduly contribute to gain in performance. The high wing supports has been enlarged but by reducing the wing sections at the root a "roll-over" effect has been produced which in placing the eye and when other certain aerodynamic benefits. The wing span has been reduced somewhat as compared to previous models, and as indicated above, leading edge flaps have been incorporated between the ailerons and the fuselage. The flaps permit a reduction of gliding speeds over a wide range, and permit short landings in small fields over relatively high obstructions.

The landing gear is of the so-called "jumper" type with shock absorbers, shock absorbers, and shock absorbers. Air wheels equipped with dual-brake hydraulic brakes are standard. Mudguards are replaceable fairs, "bars" are optional.

Behind the cabin is a large baggage compartment which may be reached either from inside or outside. When the outside door is open, a light in the rear partition is automatically turned on. A battery meter is visible in the door.

The new Rotax with its 225 hp. 15-cylinder engine has a top speed of 135 m.p.h., cruise at 120 m.p.h. The cost of maintenance is down on the new four-cylinder. The weights are as follows: empty 1,158 lb., gross 1,275 lb. The total load limit is 1,170 lb. These figures are 815 lb. 80 m.p.h. at 200 ft. 4 gal. of oil 30 ft. baggage allowance 65 lb.

## A New Plane and a New Engine by Kinner

THEir PART five weeks have witnessed the launching of two new products from the Kinner plant at Glendale, Cal.—the first, an airplane, the "Sporting" the first of a series of low-cost monoplanes to receive Department of Commerce approval and the second, a new engine in several important respects departs from the small Kinner tradition.

The "Sporting" is similar in general appearance to its predecessors, but by changing the wing tips somewhat and shaping up the power, its performance and maneuverability have been improved. When the older model landed at top speed of 110 m.p.h. the new machine has been skidded at 135. It



Latest Kinner piston plant the C-1 at 245 hp.

cranks sufficient fuel to ensure for four hours of 120 m.p.h. The power plant is a model B Kinner engine of 125 hp.

The welded steel tube fuselage is joined out to an oval section and fabric covered. The wing ribs which carry the leading gear and form points of attachment for the cabined wing panels are also an oval steel tubing and built integral with the fuselage frame. The wing panels are built up an oval space over each steel tube compression joint and double door hinging. This set in oval and the whole is fabric covered. Ailerons are also of wood and are of the tapered chord type incorporating friction ball-and differential control.

Tail surfaces are of solid steel joined internally with aluminum brack. Stabilizer is adjustable into the cockpit. The landing gear is the conventional divided type. The shock strut is Kinner-built. Aileron hinges are Timber roller bearings are regular equipment. A tail wheel is provided.

The engine is arranged for side-by-side seating with dual control. A water-cooled pump may be fitted in either end. Large baggage compartments are available in each wing stub. The main engine and magnetron equipment is furnished and motor propeller is standard equipment.

General specifications are: span, 34 ft. 3 in., length overall, 24 ft. 10 in., height, 7 ft. 2 in., wing area (standard ailerons) 174.23 sq. ft. weight empty 1,170 lb. payload 225 lb. gross weight

1,675 lb., wing loading, 12.25 lb. per sq. ft., power loading, 14.0 lb. per hp.

**TURBOSUP** in the engine, we find a similar variation from Kinner tradition as established by the well-known B, E, and C series. Hereafter the Kinner ailerons have been found to vibrate against with five cylinders only. In the new C-2, two extra cylinders have been added and the horsepower rating boosted to 240. For the first time this power installation system has been installed. On the 240-hp model an 11-in. impeller pressure in engine speed is obtained but remains the same as in the C-1. By increasing the necessary parts, applying a General Electric type impeller and changing the cylinder it will be possible to reproduce the engine in approximately 420 hp, at 1,800 r.p.m. and 300 hp at 2,200 r.p.m.

The cylinders and pistons in the C-2 are very similar to those employed in the C-3 model. Some changes have been made in the timing however, and the motor has bearings have been last changed with the head to reduce the possibility of oil leakage. The motor arms have roller bearing sides and are so designed that lubrication of the motor arms must be done from the top of the top of the path.

The crankshaft is of the two-piece type which makes the use of a separate master rod possible. The shaft is started on three main bearings at the rear end and the two ends are similar to those used in the C-3 engine.

The valve mechanism differs also from that at previous Kinner models in which each pair of valves has been operated from a short individual camshaft set in the overhead and graduated from the overhead. The C-2 is equipped with a three valve cam which runs at one-inch engine speed and is mounted in the rear half of the overhead just ahead of the slower section.

The overhead is built in two parts, sections, the power section in the front and the overhead and necessary service in the rear. The front section is made, takes care of the crankshaft bearings,

## AVIATION

February, 1939

## AVIATION

February, 1939



As the artist sees the new Fokker F-22.



The new Landing Commander in action.

the cylinders, etc., and the rear section provides for the supercharger and accessories. The new service is equipped with a valve mechanism for operating hydraulic controls with propellers. The accessory drives (which include provision for two magnetos, electric starter, generator, fuel drive pump, vacuum pump, air pressure pump, two gas synchronizers and one tachometer) are driven from an extension of the crankshaft behind the supercharger section. A Stromberg NA-15A carburetor and Stromberg magnetos are standard equipment. The oil system is of the meter pressure type with pumps and pump unit lubricated by splash in the customary manner. It has been last set with an eye toward the elimination of long oil passages in the overhead. The strongest pump is of the three-throw type and pressure is made in the pressure pump for a C-20 oil filter.

The reduction has been designed for the mounting of N.A.C.A. or other carburetors. The meter is arranged that magnetos can be replaced with

heavy spring return. Heavy springing is optional.

The general specifications are as follows: rated horsepower, 200 at 1,800 r.p.m., number of cylinders, 7; bore, 5 1/2 in.; stroke, 6 in.; displacement, 1,044 cu. in.; compression ratio, 5.25 to 1; exhaust 3 in.; 30 S.A.E. standard dry weight (without carburetor, air filter, exhaust collector ring, starter or propeller hub but including battery) approximately 775 lb.; overall diameter 21 1/2 in.; overall length, 64 in.; mounting bolt circle, 20-in. diameter.

## Lanning's Newest Commander

FOR the first time in more than 20 years of building airplanes for other people, General Lanning has turned out a new design primarily for his own personal use. He and Mrs. Lanning, who is a social pilot, will use the new Commander as a means of commuting and for pleasure flying.

The new ship is a direct descendant of the well known Lanning Commander biplane, but it shows some 40 mph. increase in top speed for the same landing speed. The cost of a complete wing system with upper trailing edge flap attachment internally in this improved performance.

The control features are apparent from the accompanying three-view drawing. Seats are provided for four people in the cabin. The ship weighs 1,450 lb. empty, maximum 1,800 lb. with load. The wing span is 42 ft. and the

area 280 sq. ft. The ailerons are of the split-type controlled by the National Advisory Committee for Aeronautics with relatively short span and long chord, insuring desired differential action much higher than those heretofore used. The power plant is a Wright J6 Mustang of 200 hp.

## The F-22 Fokker Transport

**A**LTITUDE the ship is still under construction and complete details are lacking in the aircraft, we are able to show herewith a preview of the Fokker F-22 airplane which has been ordered for European and Colonial service by both K.L.M. and A.R. Netherlands. The machine is designed for a gross weight of 26,000 lb. to include 22 passengers, a crew of four, and a total baggage and mail cargo of 1,200 lb. Total Pratt & Whitney 1,100 hp. engines of 525 hp. each mounted in the leading edge of the wing, furnish the power. The approximate main dimensions of the ship are given as 120 ft. span and 71 ft. length overall. Fuel capacity is in the neighborhood of 760 U. S. gal. and the design has been calculated to maintain 6,200 ft. altitude with full load and any one engine out of the running.

The seating arrangement is novel. The first point sits well forward in the nose where he has a very excellent view all around. The second pilot is located behind and to the right, and controls from the left the radio operator will be facing aft. Our passenger compartment is located under the nose window toward the tail. Behind the cockpit is a compartment for the crew, and the cabin proper is divided into two compartments with the usual windows and window equipment. The landing gear is not retracting but it will retract.

## Power 50, Altitude Record Holder

THE POPEYE 50 T.O.P. personal purpose military is a four-seater light on many a military airfield in Europe



The Kinner Sporting is now in service.



The special Popeye 50 is a four-seater light on many a military airfield in Europe.



## THE BUYERS' LOG BOOK

## AVIATION's Card Index of New Equipment

This department is equipped to help readers locate manufacturers of new parts, accessories or materials

## ACCESSORIES

## Camera, aerial

Fairchild Aerial Camera Corporation,  
Windsor, L. I., N. Y.

**A** COMPACT, low priced (under \$200) aerial camera is being distributed by Fairchild. Features are 3½ in. variable tension leaf plate shutter, 1/10 to 1/1000 sec. Lens 74.8 of Tessar type with 98-in. focal length 13-in. film pack, cut 35 in. plates. Filter apparatus available. Two hand grips. Dimensions (except for handles) 6½ x 5½ in. Weight under 15 lb. Suitable for ground use.

Aviation, February, 1938

## ENGINE ACCESSORIES

## Gas pressure

Standard Oil Company of New Jersey, Inc.,  
New York, N. Y.

**E**QUIPMENT for starting airplane engines in cold weather by direct injection of propane gas into the intake manifold offered, including portable tank, reducing valve, hose and injection "gun." Simple bypass arrangement permanently fixed to engine manifold for quick connection. Self contained units for permanent installation on airplanes also available.

Aviation, February, 1938

## INSTRUMENTS

## Catalog

Kollman Instrument Company,  
Brooklyn, N. Y.

**NEW** catalog "Kollman Precision Aircraft Instruments" evolved containing an account of the company's background, a description of its manufacturing and testing facilities, and a complete and detailed listing of its products. Weight schedule and installation diagrams are given for each instrument. Laminated form, in heavy paper covers. The catalog is available on request.

Aviation, February, 1938

## PARTS

## Bearings (Camels)

The Timken Roller Bearing Company,  
Canton, Ohio

**A** REVISED edition of the Timken Engineering Journal is now available. In loose-leaf form in a substantial 3-ring binder, the book contains some 230 pages of technical and engineering information on bearing problems in general, and also includes complete descriptions of the various Timken products together with tabulated data on ratings, load calculation, selection and maintenance.

Aviation, February, 1938

## RADIO

## Aeroplane receiver

Westport Manufacturing Company,  
Glenview, Ill.

**WESTPORT** Junior Model AR-60 designed specially for private airplane installations is a compact 4-tube superheterodyne receiver receiving weather, radio range and communication frequency bands (200-600 kc. and 140-1500 kc. available by selector switch). Direct operation, weighs 4½ lb., complete, cost—\$49.95 net. Operates from 12 v. "A" and 144 v. "B" batteries.

Aviation, February, 1938

## RADIO

## Airplane receiver

RCA Victor Company, Inc.,  
Columbia, N. J.

**A** NEW eight-tube superheterodyne radio receiver has been announced which covers a frequency range from 150 to 15,000 kc. in five bands. Always weather, aviation communication, radio range and foreign and domestic communication broadcasts can be heard. Self contained with built in loudspeaker, streamlined airplane type dash. For 110-220-volt 60 cycle current. 21 inch x 14 inch x 10 inch.

Aviation, February, 1938

## RADIO

## Direction finder

Washington Institute of Technology,  
Washington, D. C.

**DIRECT-AIR** Model 23-A direction finder developed for military reception for private owners and sportsman pilots. Secret operation from any aircraft under no secret. Loop antenna used. Provides guide to fly direct course to city, commercial, government or marine radio stations that can be heard. Dist. indicated "on course" position. Dimensions of receiver unit—14½ x 14½ x 10 in.

Aviation, February, 1938

## SHOP EQUIPMENT

## Paint spray gun

Electrical Painting Equipment Company, Inc.,  
247 Park Avenue, New York, N. Y.

**THE** Name-Airless Painting Machine uses compressed air to atomize paint and other liquids instead of compressed air. Prints to a shiny line without producing paint fog. Self-contained, hand-held unit incorporates paint supply, electric motor and spray control. Operates from 110 volt (220-volt machines special). AC or DC power supply. Weighs 5½ lb. and may be packed in small space.

Aviation, February, 1938



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That is why you owe it to your future as a pilot, maintenance man or executive in aviation to complete flying trainings before you choose any! Ask any airmen pilot for his opinion and, meantime, send for an Illustrated Boeing Bulletin. It gives entrance requirements, courses, costs, living conditions at the Oakland Airport, et cetera. The coupon brings it.

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478 Folsom Highway, Newark, New Jersey

**WESTON**  
Instruments



## WESTERN AIR EXPRESS picks SOCONY-VACUUM



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MOBILGREASE FOR ROCKER-ARM LUBRICATION



For Rocker Arms? Pick a double-range lubricant.

On the Rocky Mountain Division of Western Air Express—where temperature and altitude conditions would play havoc with ordinary greases—they have standardized on Mobilgrease—the double-range lubricant.

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oil Aero-Oils and Aviation Mobilgas—both double-range products. These gasoline and oils have been developed to give maximum power and protection at any altitude—at any speed—at any temperature.

That is why so many commercial lines choose them. Socony-Vacuum double-range products include—Mobilol Aero Oil, Mobilgrease, Aviation Mobilgas, Socony Vacuum Instrument Oil and Socony-Vacuum Compass Fluid.

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Here it is...an aviation product with the dependability so necessary to all items of aviation service—because—it was developed specifically for its purposes.



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This receiver is a completely self-contained, AC operated, superheterodyne, bank of the finest mass coils by skilled craftsmen. Its high quality, performance and dependability really place it well above the class of standard obtainable with ordinary receivers. You are invited to write for full details. Address: Airways Radio Section, RCA Victor Company, Inc., Camden, N. J., or Air Associates, Inc.

AVIATION RADIO SECTION

RCA Victor Co., Inc.

CAMDEN, N. J.

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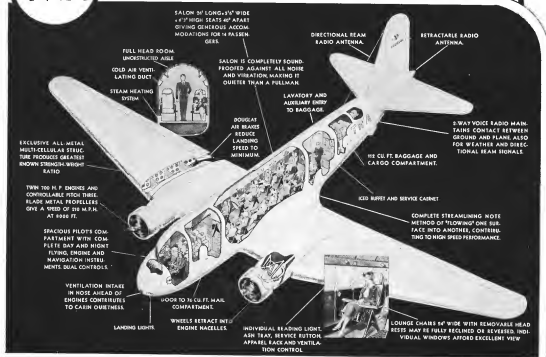


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